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AVIA 13/734

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AEROPLANE AND AIRCRAFT EXPERIMENTAL ESTABLISHMENT.

BOSCOMBE DOWN.

UNCLASSIFIED

TS. 10/7/63

Kittyhawk II.F.L.220.  
(Merlin V.1650-1.)

STOCK

9

Climb and level speed performance and position

REDUCE TO 12

99 AUTHORIZED

DATE 6.1.53.

A.&A.E.E.ref:- 4484/1-A.S.76/4

M.A.P.ref:- R.A.1862/D.A.N.A.1.

This report deals with the aircraft (or equipment) as tested. Action to remedy defects or decisions to accept items not in strict compliance with the specification are matters for decision and action by the M.A.P.

Progress of issue of report.

| Report No.                 | Title.   |
|----------------------------|--|
| 1st Part of A.A.E.E./783a. | F.L.220 - Weights and loading data.                                  |
| 2nd " " "                  | F.L.220 - Flame damping trials with short stub exhaust flame damper. |
| 3rd " " "                  | F.L.220 - Carbon monoxide contamination tests.                       |

SUMMARY.

Climb and maximum level speed performance has been measured on Kittyhawk II. F.L.220. Briefly, the results were :-

Max. rate of climb at full throttle height M.S. gear 2020 ft/min @ 10,200 ft.  
" " " " " " " " F.S. " 1620 ft/min @ 17,000 ft.

Time to reach 10,000 ft. 5.0 mins.  
" " " 20,000 ft. 10.9 "  
" " " 30,000 ft. 21.8 "

Service ceiling. 34,300 feet.  
Estimated absolute ceiling. 35,400 feet.

Max. level true air speed in M.S. supercharger gear 347 m.p.h. @ 14,700 ft.  
" " " " " " " F.S. " 354 " @ 20,400 ft.

1. Introduction.

Level speed and climb trials have been carried out on Kittyhawk II.F.L.220, which is fitted with a Packard built Merlin V.1650-1 engine and a Curtiss Electric Propeller.

The tests were carried out in August, 1942. Preliminary results have already been forwarded to the Ministry of Aircraft Production in our letter ref. A.&A.E.E./4484/1-A.S.76/4 dated 3rd September, 1942.

2. Scope of tests.

The climbs were made at the climbing speed recommended by the manufacturers of 160 m.p.h.A.S.I. to 20,000 feet, reducing speed by 2 m.p.h. per 1000 ft. thereafter. The change from M.S. to F.S. supercharger gear was made at 13,000 ft. and the engine speed increased from 2850 r.p.m. to 3000 r.p.m. at 20,000 feet. The gills were fully open during the climb.

The position error correction was measured by the aneroid method. Level speeds were measured in M.S. and F.S. supercharger gears.

3. Condition of aircraft relevant to tests.

The armament of this aircraft consisted of six 0.5" machine guns, three in each wing. The muzzles protruded about three inches from the leading edge and were tape-bound, but the ejector chutes under the wings were open.



Aerials stretched from the fin to the wing tips and to the rear of the cockpit. I.F.F. aerials were also fitted. There was no aerial mast.

An external mirror was mounted over the cockpit and slightly to port. Under the fuselage were fittings for an external petrol tank or bomb. There were also slots in the under surface of the wing for bomb racks. There was a landing lamp in the port wing. The aircraft was tested throughout at 8910 lb. weight.

Engine limitations for the Merlin V.1650-1 used on the tests were:-

|   | R.P.M. | Boost.<br>(ins. of Hg). |
|---|--------|-------------------------|
| Maximum for take-off,                             | 3000   | 54½                     |
| Maximum for all-out level flt.<br>(5 min. limit). | 3000   | 48                      |
| Maximum on climb (below 20,000 ft)                | 2850   | 48                      |
| Maximum for continuous<br>(above 20,000 ft)       | 3000   | 48                      |
| cruising rich...                                  | 2650   | 44                      |
| Maximum for continuous<br>cruising weak...        | 2650   | 38                      |

The propeller was a Curtiss Electric, three-bladed right handed type of 11' diameter with metal blades, number 32236.

#### 4. Results.

The results of the climb trials are given in Table I and in figure 1, and of the level speed trials in Table II and figure 2.

The position of the pitot head is detailed in figure 3, the position error correction in figure 4, and the altimeter correction when corrected to the static of the pressure head in figure 5.

The results have been summarised on the first page of this report.

#### 5. Discussion.

The results given are based on the  $p_2^{1/2} \sigma^{1/2}$  correction method. When correction is by the method detailed in A.&A.E.E. Memorandum dated 27.8.42., the following differences in performance are found:-

Maximum rate of climb at full throttle height M.S. gear increased by 70 ft/min. to 2090 ft/min. at 10,400 feet.

Maximum rate of climb at full throttle height F.S. gear increased by 50 ft/min. to 1670 ft/min. at 17,100 ft.

Maximum level true air speed in M.S. supercharger gear increased by 6 m.p.h. to 353 m.p.h. at 15,000 ft.

Maximum level true air speed in F.S. supercharger gear increased by 4 m.p.h. to 358 m.p.h. at 20,400 ft.



• TABLE I.

Performance on Climb.  
Gills fully open.

| Standard height (ft) | Rate of climb (ft/min) | Time (mins) | T.A.S. m. p. h. | A.S.I. m. p. h. | Corrections m. p. h. |       | R.P.M. | Boost Ins. of Hg. | Supercharger gear. |
|----------------------|------------------------|-------------|-----------------|-----------------|----------------------|-------|--------|-------------------|--------------------|
|                      |                        |             |                 |                 | P.E.                 | Comp. |        |                   |                    |
| 2000                 | 2020                   | 1.0         | 168             | 160             | +3.7                 | -0.1  | 2850   | 48.0              | M.S.               |
| 4000                 | ↓                      | 2.0         | 174             | ↓               | ↓                    | -0.2  | ↓      | ↓                 | ↓                  |
| 6000                 | ↓                      | 3.0         | 179             | ↓               | ↓                    | -0.3  | ↓      | ↓                 | ↓                  |
| 8000                 | ↓                      | 4.0         | 184             | ↓               | ↓                    | -0.4  | ↓      | ↓                 | ↓                  |
| 10000                | ↓                      | 5.0         | 190             | ↓               | ↓                    | -0.5  | ↓      | ↓                 | ↓                  |
| 10200*               | ↓                      | 5.05        | 191             | ↓               | ↓                    | -0.5  | ↓      | ↓                 | ↓                  |
| 12000                | 1830                   | 6.0         | 196             | ↓               | ↓                    | -0.6  | ↓      | 45.2              | ↓                  |
| 14000                | 1620                   | 7.2         | 202             | ↓               | ↓                    | -0.7  | ↓      | 48.0              | F.S.               |
| 16000                | ↓                      | 8.3         | 209             | ↓               | ↓                    | -0.8  | ↓      | ↓                 | ↓                  |
| 17000 <sup>δ</sup>   | ↓                      | 9.0         | 212             | ↓               | ↓                    | -0.9  | ↓      | ↓                 | ↓                  |
| 18000                | 1530                   | 9.6         | 215             | ↓               | ↓                    | -1.0  | ↓      | 46.2              | ↓                  |
| 20000 <sup>δ</sup>   | (1350)                 | 10.9        | 220             | 158             | +3.6                 | -1.0  | ↓      | 42.7              | ↓                  |
|                      | (1500)                 |             |                 |                 |                      |       | 2990   | 44.7              | ↓                  |
| 22000                | 1300                   | 12.5        | 222             | 154             | +3.5                 | -1.1  | ↓      | 41.7              | ↓                  |
| 24000                | 1100                   | 14.2        | 223             | 150             | +3.3                 | -1.2  | ↓      | 38.8              | ↓                  |
| 26000                | 910                    | 16.1        | 225             | 146             | +3.2                 | -1.2  | ↓      | 36.0              | ↓                  |
| 28000                | 720                    | 18.5        | 227             | 142             | +3.0                 | -1.3  | ↓      | 33.2              | ↓                  |
| 30000                | 520                    | 21.8        | 228             | 138             | +2.8                 | -1.3  | ↓      | 30.7              | ↓                  |
| 32000                | 320                    | 26.7        | 230             | 134             | +2.7                 | -1.4  | ↓      | 28.0              | ↓                  |
| 34000                | 130                    | 35.8        | 231             | 130             | +2.5                 | -1.5  | ↓      | 25.5              | ↓                  |

\* Full throttle height M.S. supercharger gear.

<sup>δ</sup> " " " F.S. " "

<sup>δ</sup> R.P.M. increased.

Service ceiling ... 34,300 ft.

Estimated absolute ceiling .. 35,400 ft.

TABLE II.

Level speeds and boost at height.  
Gills in neutral position.

| Standard height (ft) | True Airspeed m. p. h. | A.S.I. m. p. h. | Position error correction | Compressibility correction | R.P.M. | Boost ins. of Hg. | Supercharger gear. |
|----------------------|------------------------|-----------------|---------------------------|----------------------------|--------|-------------------|--------------------|
| 8,000                | 320                    | 277             | +7.9                      | -1.5                       | 3000   | 48.1              | M.S.               |
| 10,000               | 328.5                  | 276             | +7.9                      | -1.8                       | ↓      | 48.1              | ↓                  |
| 12,000               | 336                    | 274             | +7.9                      | -2.3                       | ↓      | 48.1              | ↓                  |
| 14,000               | 344.5                  | 272.5           | +7.8                      | -2.8                       | ↓      | 48.1              | ↓                  |
| 14,700 *             | 347                    | 271             | +7.8                      | -3.0                       | ↓      | 48.1              | ↓                  |
| 16,000               | 346                    | 265             | +7.6                      | -3.1                       | ↓      | 46.1              | ↓                  |
| 18,000               | 342                    | 254             | +7.2                      | -3.2                       | ↓      | 42.6              | ↓                  |
| 20,000               | 337                    | 242.5           | +6.8                      | -3.3                       | ↓      | 39.3              | ↓                  |
| 16,000               | 336                    | 257.5           | +7.3                      | -3.0                       | 2980   | 48.0              | F.S.               |
| 18,000               | 344                    | 255.5           | +7.2                      | -3.4                       | ↓      | 48.0              | ↓                  |
| 20,000               | 352.5                  | 254             | +7.2                      | -3.7                       | ↓      | 48.0              | ↓                  |
| 20,400 <sup>δ</sup>  | 354                    | 253.5           | +7.2                      | -3.8                       | ↓      | 48.0              | ↓                  |
| 22,000               | 352                    | 245.5           | +6.9                      | -3.9                       | ↓      | 45.3              | ↓                  |
| 24,000               | 349                    | 235.5           | +6.4                      | -3.9                       | ↓      | 41.8              | ↓                  |
| 26,000               | 344.5                  | 224.5           | +6.1                      | -3.9                       | ↓      | 38.3              | ↓                  |

\* Full throttle height M.S. supercharger gear.

<sup>δ</sup> " " " F.S. " "



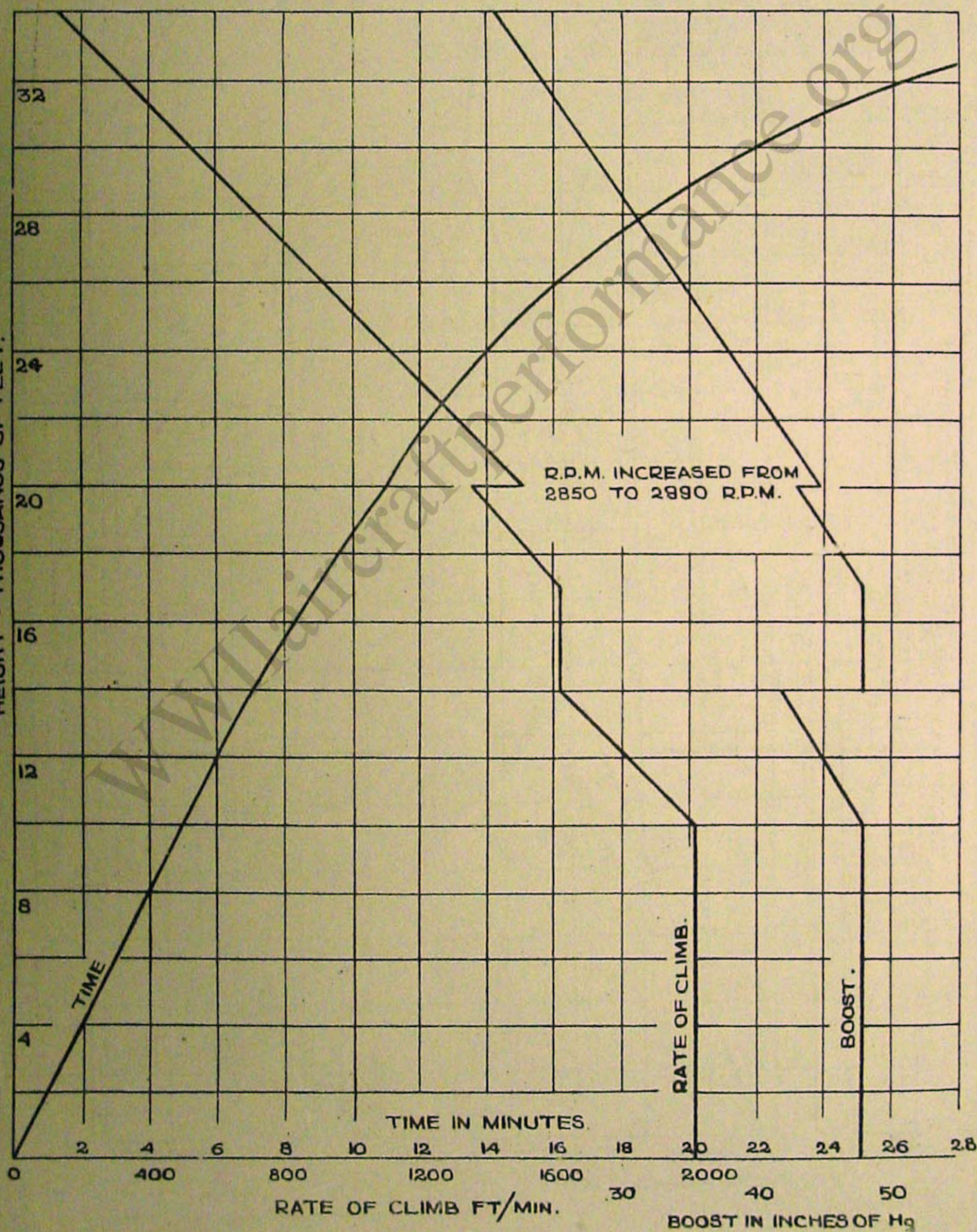
FIG. 1

# KITTYHAWK II FL-220

WEIGHT ~ 8910 lbs.

## PERFORMANCE ON CLIMB.

GILLS FULLY OPEN.





# KITTYHAWK II FL-220

FIG 2

WEIGHT ~ 8910 lb.

LEVEL SPEEDS AND BOOST AT HEIGHTS.

GILLS IN NEUTRAL POSITION.

