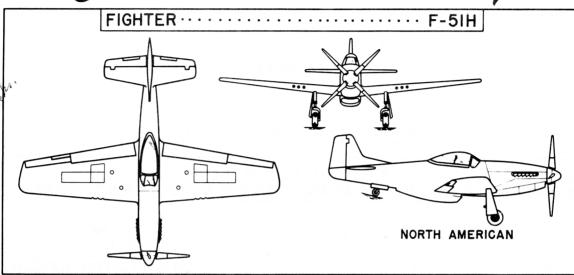


Characteristics Summary



Wing area236.0 sq ft

Length 33.3 ft

Span 37.0 ft

Height 13.7 ft

AVAILABILITY Number available			PROCUREMENT						
			Number to be delivered in fiscal years						
ACTIVE	RESERVE	TOTAL							

STATUS

SERVICE

1. Design initiated: 4 May 1940 (XF-51)

2. First acceptance: January 1945

3. Production completed: November 1945

POWER PLANT

(1) V-1650-9 Packard BHP - RPM - ALT.**

T.O: 1335 - 3000 - S.L. Max: *2220 - 3000 - 9,000 *1790 - 3000 - 22,700

Nor: 1085 - 2700 - 21,800 930 - 2700 - 31,800

*Fluid injection

** Level flight critical altitude (incl.ram)

FEATURES

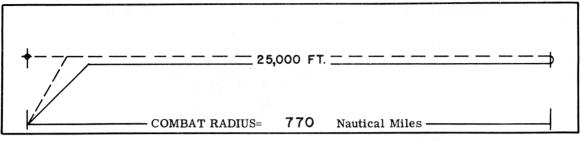
Crew: 1
Tail Warning Radar
Cockpit Heating
Max Fuel Capacity: 590 gal

ARMAMENT

Turrets: None
Guns: (6) .50 cal
Ammunition: 1820 rds
Max Bomb Load: (2) 1000 lb
Max Bomb Size: 1000 lb
Rocket Provisions:

(10) 5" HVAR

Characteristics Summary Basic Mission ····· F-51H



PERFORMANCE									
COMBAT RADIUS	COMBAT RANGE	COMBAT SPEED							
770 naut. mi	1665 naut. mi	406 knots at 25,000 ft alt, max power							
with 0 lb payload	with 0 lb payload	MAXIMUM SPEED							
at 249 knots avg. in 6.7 hours.	at 250 knots avg. in 6.8 hours.	4 O knots at 22,700 ft alt, max power							
CLIMB	CEILING	TAKE-OFF							
sea level, take-off weight normal power	32,400 ft take-off weight normal power	ground run 1480 ft ft no assist assisted							
5000 fpm sea level, combat weight maximum power	39,000 ft 500 fpm, combat weight maximum power	over 50 ft height 2490 ft no assist assisted ft							
L O A D	WEIGHTS	STALLING SPEED							
Bombs: None Ammunition: 1820 rds/.50 cal	Empty 6551 lb Combat 9430 lb	92 knots flaps down, take-off weight							
Fuel: 480 gal protected 54 % droppable 46 % external 46 %	Take - off 11,029 lb limited by mission	6.6 min. S.L. to 25,000 ft. Combat Wt., Max Power							

NOTES

- 1. PERFORMANCE BASIS: NACA standard conditions no wind, single airplane; (a) Fuel consumption based on flight tests, increased 5%
- ${f 2.}$ Major differences from F-51D: Engine and propeller changes, longer fuselage and major structural changes.
- 3. REVISION BASIS: To present data in complete agreement with Standard Aircraft Characteristics Charts.